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THESIS

INVESTIGATION OF SHEAR LAG FACTORS ON BOLTED HOT-ROLLED STEEL ANGLES

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A Thesis Submitted in Partial Fulfillment of the Requirements for the Degree of Master of Engineering (Civil Engineering) Graduate School, Kasetsart University 2013 Tawan Khankruer 2013: Investigation of Shear Lag Factors on Bolted Hotrolled Steel Angles. Master of Engineering (Civil Engineering), Major Field: Civil Engineering, Department of Civil Engineering. Thesis Advisor: Associate Professor Trakool Aramraks, Ph.D. 52 pages.

The objective of this experimental program is to investigate a shear lag factor "U" on tension members with net section fracture of bolted hot-rolled steel angle. The parameters considered in this experimental program were number of bolts, connection length (center of first hole to center of last hole), size/cross-section of angle member and ultimate tensile strength of steel angle.

The results showed that shear lag factor "U" were affected by number of bolts, connection length, size/cross section of steel angles and ultimate tensile strength of material. The shear lag factor obtained from the experiment were within the range between the values provided by AISC 2010, case 2 to case 8 depended on the values of ultimate tensile strength (F_u). Choosing the smallest value of F_u was always safe and recommended for designer. Further study should be focused on how to choose F_u for economical design.

Student's signature	Thesis Advisor's signature	/ _	/

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I would like to show my grateful thanks and am deeply indebted to Associate Professor Trakool Aramraks, Ph.D. my thesis advisor for advice, encouragement and valuable suggestions since the beginning of my chosen topic, possibility study to start the test, recommendation of solving the problem during the period of test and the complete writing of this thesis.

I especially appreciate my parents for their support in my life, teachers, professors and all my relatives and my friends, who have given me support and encouragement throughout my graduate career. Their love and companionship have been an essential source of strength to me during this course

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LIST OF ABBREVIATIONS

Notation and definitions of the terms are

 A_e = effective net area

 $A_g = gross area$

 A_n = net area

 F_u = ultimate tensile strength

 F_y = ultimate yield strength

 K_1 = an attempt to account for the ductility of material

 K_2 = the fabrication factor

 K_3 = the geometry factor

 K_4 = shear lag factor from base on research by Munse and Chesson (1963)

L = connection length

 P_n = nominal tensile strength

P_{exp}= ultimate tensile strength

Q = percent reduction in the area at rupture of a standard tensile test coupon (51 mm.gage length)

U = shear lag factor

 \tilde{x} = distance from shear plane to the center of gravity of the material connection to the shear plane

INVESTIGATION OF SHEAR LAG FACTORS ON BOLTED HOT-ROLLED STEEL ANGLES

INTRODUCTION

Nowadays, Steel Structure is used for many types of structures such as bridges, building, etc. The advantages of steel structure compared to others are light weight, homogeneous material, fast and easy erection and installation, disassembly and recycling. In most steel constructions in Thailand, hot-rolled sections are the main steel members which are usually used and member shape such as angles are often used as tension members. Bolted or fastener connection is the usual way for steel members, but when axial load is transferred to cross-section, a non-uniformed stress is produced and distributed to the net section area (Figure 1).

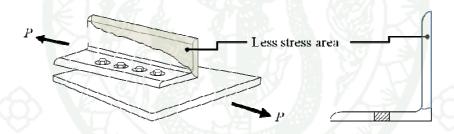


Figure 1 Non-uniformed stress distribution on the net area section of angle member.

Source: Jiravacharadet (2005)

American Institute of Steel Construction (AISC, 2010) defined effective net area of tension member which shall be determined as follows;

$$A_{e} = A_{n}U \tag{1}$$

Notation and definitions of the terms are;

 A_e = effective net area, (mm²)

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 $A_n = \text{net area, } (mm^2)$

U = the shear lag factor.

Shear lag factor was suggested by AISC, 2010 as shown below.

$$U = 1 - \left(\frac{\tilde{x}}{L}\right) \tag{2}$$

Notation and definitions of the terms are;

 \tilde{x} = distance from shear plane to the center of gravity of the material connected to the shear plane, mm.

L = connection length, mm.

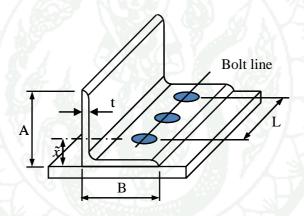


Figure 2 Typical hot-rolled angle for a bolted connection.

Source: De Paula *et al.* (2008).

OBJECTIVES

The objective of this experimental program is to investigate a shear lag factor "U" on tension members with net section fracture of bolted hot-rolled steel angle. Various parameters of the experimental program are considered such as, number of bolts, connection length (center of first hole to center of last hole), size/cross-section of angle member and ultimate tensile strength of steel angle.

The objective is summarized as follows;

- 1. To investigate behaviors of bolted hot-rolled steel angle under tension force.
- 2. To compare shear lag factor between test and design code.

Scope of Study

- 1. Parameters to be studied are as follows:
 - Number of bolts
 - Connection length (center of first hole to center of last hole)
 - Size/Cross-section of angle member
 - Difference of ultimate tensile strength
- 2. 26 numbers of single angle specimens are tested.
- 3. The test considers net section failure mode only.

- 4. All single angle specimens follow the JIS G3101 standard Grade SS400 whose mechanical properties are;
 - Yield strength of 245 MPa.
 - Ultimate tensile strength of 400-510 MPa.
 - 5. High-strength hex bolts ASTM A325M 8S, thread on shear plane
 - 6. Types of Bolted Connections are bearing type connection,

LITERATURE REVIEW

1. Previous Studies on shear Lag of Tension Members

There are three principal failure modes of members under tension forces;

- (I) yielding of the gross section,
- (II) net section failure, and
- (III) failure of the end connection.

Shear lag is related to mode (II). Such failure describes behavior at an end connection of a tension member where some but not all of the cross-sectional elements are connected. In other words, the area that is effective in resisting tension may be less than the full calculated net area (Albert, 1996).

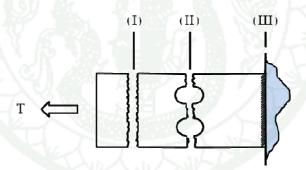


Figure 3 Three principle failure modes of tension member. (Albert, 1996). .

Most steel structure design provisions, including American Institute of Steel Construction (AISC, 2010) are based on research by Munse and Chesson (1963). Munse and Chesson found that the value of section efficiency of tension members, bolted or riveted at end connections was a function of large number of factors. This equation was developed from the results of more than 1000 tests. They summarized the experiments and proposed an equation as shown below.

$$\mathbf{A}_{\mathbf{e}} = \mathbf{K}_{1} \mathbf{K}_{2} \mathbf{K}_{3} \mathbf{K}_{4} \mathbf{A}_{\mathbf{n}} \tag{3}$$

Notation and definitions of the terms are

$$K_1 \ = \ 0.82 + 0.0032Q < 1$$

 $K_2 = 0.85$ for members with punched holes, 1.0 for members with drilled holes

$$K_3 = 1.6 - 0.7 \left(\frac{A_n}{A_g}\right)$$

$$K_4 = 1 - \left(\frac{\tilde{x}}{L}\right)$$

 A_g = gross area of cross-section

Q = percent reduction in the area at rupture of a standard tensile test coupon (51 mm gauge length)

 \tilde{x} = distance from shear plane to the center of gravity of the material connected to the shear plane, mm.

L = connection length, mm.

The factor K_1 is an attempt to account for the ductility of material. The term Q applied in definition of K_1 is a reduction to the area of a standard tensile test coupon (51 mm gauge length). The fabrication factor K_2 , is used to account for the reduction in efficiency due to the effect of punching the holes. To account for the effect of hole spacing on the connection, or the g/d ratio, a geometry factor, K_3 , is included. Finally, K_4 is the shear lag factor, which takes into account both the eccentricity in the connected parts and the connection length.

In expression for K_4 , \tilde{x} refers to the distance from the face of gusset plate to the center of gravity of member and L is connection length, taken as the distance between extreme fasteners. Munse and Chesson note that Equation 3 predicted the net section efficiency within \pm 10 percent range for most of test data.

For most practical case the ductility factor (K_1) and geometric factor (K_3) can be taken as unity (Wu and Kulak, 1993). The fabrication factor K_2 can be given the value 0.85 for punched holes and 1.0 for drilled holes (Munse and Chesson, 1963). According to these assumptions, the Equation 3 can be written as

$$A_{e} = \left(1 - \frac{\tilde{x}}{L}\right) A_{n} \tag{4}$$

According to the Munse and Chesson (1963) formulation by substituted effective net section area from Equation 4 to Equation 3, the predicted tensile load of an angle will be

$$P_{n} = \left(1 - \frac{\tilde{x}}{L}\right) A_{n} F_{u} \tag{5}$$

2. Current Design Specifications

American Institute of Steel Construction (AISC, 2010), Chapter D indicates that for the design tensile strength, $\phi_t P_n$, and the allowable tensile strength, P_n/Ω_t , of tension members shall be the lower value obtained according to the limit states of tensile yielding in the gross section and tensile rupture in the net section.

a) For tensile yielding in the gross section

$$P_{n} = F_{y}A_{g} \label{eq:phi}$$

$$\phi_{t} = 0.90 \; (LRFD) \; \text{and} \; \Omega_{t} = 1.67 \; (ASD)$$

b) For tensile rupture in the net section

$$\begin{split} P_{n} &= F_{u}A_{e} \\ \phi_{t} &= 0.75 \text{ (LRFD) and } \Omega_{t} = 2.00 \text{ (ASD)} \end{split} \label{eq:partial_partial}$$

Notation and definitions of the terms are;

 P_n = nominal tensile strength, (N)

 A_e = effective net area, (mm²)

 $A_g = gross area of member, (mm²)$

 F_y = specified min. yield stress of the type of steel being used, (MPa)

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Notation and definitions of the terms are;

 F_u = specified min. tensile strength of the type of steel being used, (MPa)

 ϕ_t = resistance factor for tension.

 Ω_t = Safety factor for tension.

To compute the net section strength of tension member in a way that includes the shear lag effect, an "effective" net area is used instead of the net area in order to take into account the reduction of efficiency as a result of this effect. The American Institute of Steel Construction (AISC, 2010), Section D3, defines effective net area of tension member shall be determined as follows:

$$A_{e} = A_{n}U \tag{8}$$

Notation and definitions of the terms are

 A_e = effective net area, (mm²)

 $A_n = \text{net area, } (mm^2)$

U =the shear lag factor.

For all tension members where the tension load is transmitted directly to each of cross-sectional elements by fasteners or welds, the shear lag factor (U) equal to 1.0 is permitted to be used. And for all tension members, except plates and HSS, where the tension load is transmitted to some but not all of the cross sectional elements by fasteners or longitudinal welds, the shear lag factor shall be calculated by

$$U = 1 - \frac{\tilde{x}}{L} \tag{9}$$

However, AISC (2010) also defines an additional value of shear lag factor for single angle member, separated into two cases;

Table 1 Shear lag factors for connections to tension members.

Case	Description	of Element	Shear Lag Factor, U
2	All tension member HSS, where the tentransmitted to some cross sectional elem longitudinal welds welds in combination welds. (Alternative HP, Case 7 may be Case 8 may be used	U = 1-x~/L	
8	Single angles (if <i>U</i> is calculated per Case 2, the large value is	with 4 or more fasteners per line in direction of loading	U = 0.80
	permitted to be used)	with 2 or 3 fasteners per line in direction of loading	U=0.60

Source: AISC (2010)

MATERIALS AND METHODS

1. Hypothesis

- 1.1 Failure of steel angle under tensile force is related to assumed crosssection area.
- 1.2 Testing steel angles has the same properties (F_y = 245MPa, F_u = 400-510 MPa) for all the batch.

2. Materials and Equipment

2.1 Materials

- 2.1.1 Hot-rolled steel angles; JIS 3101 Grade SS400
 - 2.1.1.1 Angle size 50x50x4, 500 mm long. (Figure 4)
 - 2.1.1.2 Angle size 50x50x6, 500 mm long. (Figure 4)
 - 2.1.1.3 Angle size 75x75x6, 500 mm long. (Figure 5)
- 2.1.2 Gusset plate 10 mm thickness (Figure 6)
- 2.1.3 ASTM A325M bolts, diameter 16.0 mm (Figure 7)

2.2 Equipment

- 2.2.1 Universal testing machine with monitor, capacity 2000 kN
- (Figure 9)
- 2.2.2 Data logger (Figure 10)
- 2.2.3 Dial gage

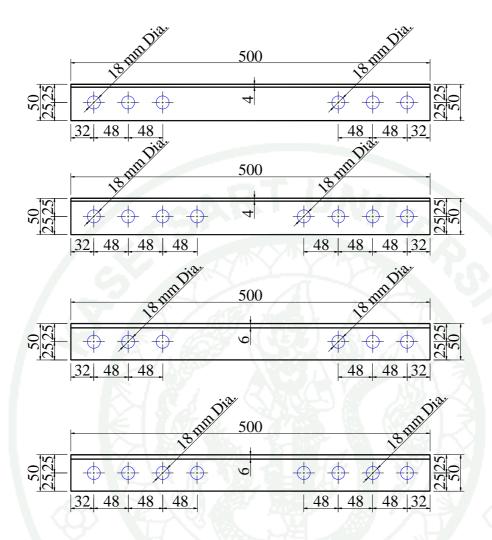


Figure 4 50x50x4 mm and 50x50x6 mm. hot-rolled steel angles

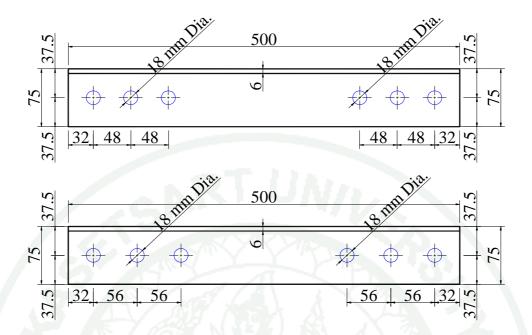


Figure 5 75x75x6 mm hot-rolled steel angles

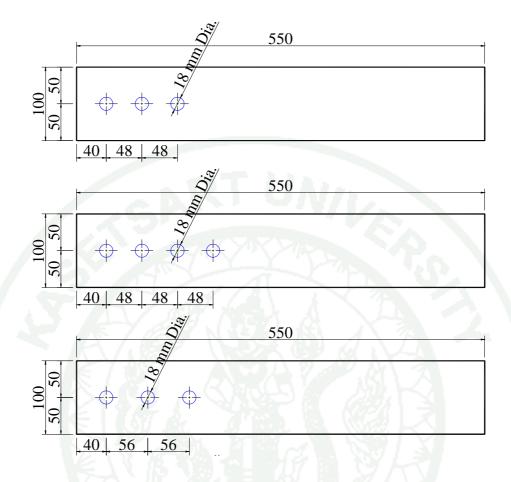


Figure 6 100x550x10 mm gusset plates.



Figure 7 ASTM A325 bolts, diameter 16.0 mm

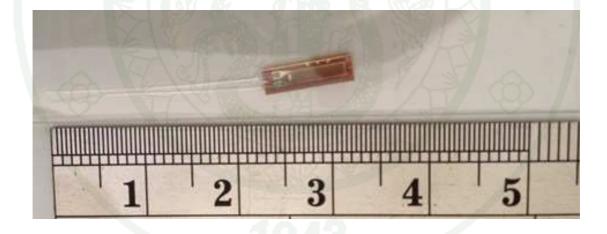


Figure 8 Strain gages (gage length 5.0 mm)



Figure 9 Universal Testing Machine (UTM) with monitor, Capacity 2000 kN.



Figure 10 Data logger.

3. Testing Procedure

3.1 General

To investigate the behavior of shear lag effect on hot-rolled single angle member, three different sizes of angles are produced. Twenty six specimens (Table 2) are tested in the laboratory. Results from the test are compared with the results calculated according to the AISC (2010). Various parameters of the experimental program are as follows:

- Number of bolts
- Connection length (center of first hole to center of last hole)
- Size/Cross-section of angle member
- Difference ultimate tensile strength

3.2 Specimens Description

The 26 specimens with the length of 500 mm. are prepared for the laboratory tests. JIS G 3101 grade SS400 is common class of steel member in Thailand whose mechanical properties are;

- Yield strength of 245 MPa
- Ultimate tensile strength of 400-510 MPa.

Angle members to be tested are varied in size and thickness. Each side of angle member has been drilled for bolted connections with 1 line pattern. Three and four bolts for a line are prepared and considered as factors of the test results. All holes are drilled to 18.0 mm diameter for ASTM A325 bolt diameter 16.0 mm. The description of the specimens and gusset plate geometry and dimensions are presented in Table 2.

Table 2 Description of specimens

Test	Group	Angle denomination	A (mm)	B (mm)	t (mm)	Number of bolt lines	No. of holes per bolt line
1		50-4-311	50	50	4	1	3
2		50-4-312	50	50	4	1	3
3	G1	50-4-313	50	50	4	1	3
4		50-4-314	50	50	4	1	3
5		50-4-315	50	50	4	1	3
6		50-4-411	50	50	4	1	4
7		50-4-412	50	50	4	1	4
8	G2	50-4-413	50	50	4	1	4
9		50-4-414	50	50	4	1	4
10		50-4-415	50	50	4	1	4
11		50-6-311	50	50	6	1	3
12		50-6-312	50	50	6	1	3
13	G3	50-6-313	50	50	6	7 1 3	3
14		50-6-314	50	50	6		3
15		50-6-315	50	50	6	1	3
16	S	50-6-411	50	50	6	1 - 1	4
17		50-6-412	50	50	6	1	4
18	G4	50-6-413	50	50	6	1	4
19		50-6-414	50	50	6	1	4
20		50-6-415	50	50	6	1	4
21	8	75-6-311	75	75	6	1	3
22	G5	75-6-312	75	75	6	1	3
23		75-6-313	75	75	6	1	3
24		75-6-314	75	75	6	1	3
25	G6	75-6-315	75	75	6	1	3
26		75-6-316	75	75	6	1	3

Table 2 (Continued)

Test Group Angle denomination (mm) d x L A An mm2 mm2 1 50-4-311 18.0 13.7 96 389.2 317.2 2 50-4-312 18.0 13.7 96 389.2 317.2 3 G1 50-4-313 18.0 13.7 96 389.2 317.2 4 50-4-314 18.0 13.7 96 389.2 317.2 5 50-4-315 18.0 13.7 96 389.2 317.2 6 50-4-411 18.0 13.7 144 389.2 317.2 7 50-4-412 18.0 13.7 144 389.2 317.2 8 G2 50-4-413 18.0 13.7 144 389.2 317.2 9 50-4-415 18.0 13.7 144 389.2 317.2 10 50-6-311 18.0 13.7 144 389.2 317.2 11 50-6-312								
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3 G1 50-4-313 18.0 13.7 96 389.2 317.2 4 50-4-314 18.0 13.7 96 389.2 317.2 5 50-4-315 18.0 13.7 96 389.2 317.2 6 50-4-411 18.0 13.7 144 389.2 317.2 7 50-4-412 18.0 13.7 144 389.2 317.2 8 G2 50-4-413 18.0 13.7 144 389.2 317.2 9 50-4-414 18.0 13.7 144 389.2 317.2 10 50-4-415 18.0 13.7 144 389.2 317.2 11 50-6-311 18.0 13.7 144 389.2 317.2 11 50-6-312 18.0 13.7 144 389.2 317.2 11 50-6-312 18.0 14.4 96 564.4 456.4 12 50-6-315 18.0	1		50-4-311	18.0	13.7	96	389.2	317.2
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5 50-4-315 18.0 13.7 96 389.2 317.2 6 50-4-411 18.0 13.7 144 389.2 317.2 7 50-4-412 18.0 13.7 144 389.2 317.2 8 G2 50-4-413 18.0 13.7 144 389.2 317.2 9 50-4-414 18.0 13.7 144 389.2 317.2 10 50-4-415 18.0 13.7 144 389.2 317.2 11 50-6-311 18.0 14.4 96 564.4 456.4 12 50-6-312 18.0 14.4 96 564.4 456.4 13 G3 50-6-313 18.0 14.4 96 564.4 456.4 14 50-6-315 18.0 14.4 96 564.4 456.4 15 50-6-411 18.0 14.4 144 564.4 456.4 16 50-6-412 18.0	3	G1	50-4-313	18.0	13.7	96	389.2	317.2
6 50-4-411 18.0 13.7 144 389.2 317.2 7 50-4-412 18.0 13.7 144 389.2 317.2 8 G2 50-4-413 18.0 13.7 144 389.2 317.2 9 50-4-414 18.0 13.7 144 389.2 317.2 10 50-4-415 18.0 13.7 144 389.2 317.2 11 50-6-311 18.0 14.4 96 564.4 456.4 12 50-6-312 18.0 14.4 96 564.4 456.4 13 G3 50-6-313 18.0 14.4 96 564.4 456.4 14 50-6-315 18.0 14.4 96 564.4 456.4 15 50-6-315 18.0 14.4 96 564.4 456.4 16 50-6-412 18.0 14.4 144 564.4 456.4 18 G4 50-6-413	4		50-4-314	18.0	13.7	96	389.2	317.2
7 50-4-412 18.0 13.7 144 389.2 317.2 8 G2 50-4-413 18.0 13.7 144 389.2 317.2 9 50-4-414 18.0 13.7 144 389.2 317.2 10 50-4-415 18.0 13.7 144 389.2 317.2 11 50-6-311 18.0 14.4 96 564.4 456.4 12 50-6-312 18.0 14.4 96 564.4 456.4 13 G3 50-6-313 18.0 14.4 96 564.4 456.4 14 50-6-314 18.0 14.4 96 564.4 456.4 15 50-6-315 18.0 14.4 96 564.4 456.4 16 50-6-411 18.0 14.4 144 564.4 456.4 17 50-6-413 18.0 14.4 144 564.4 456.4 18 G4 50-6-413	5		50-4-315	18.0	13.7	96	389.2	317.2
8 G2 50-4-413 18.0 13.7 144 389.2 317.2 9 50-4-414 18.0 13.7 144 389.2 317.2 10 50-4-415 18.0 13.7 144 389.2 317.2 11 50-6-311 18.0 14.4 96 564.4 456.4 12 50-6-312 18.0 14.4 96 564.4 456.4 13 G3 50-6-313 18.0 14.4 96 564.4 456.4 14 50-6-314 18.0 14.4 96 564.4 456.4 15 50-6-315 18.0 14.4 96 564.4 456.4 16 50-6-411 18.0 14.4 144 564.4 456.4 17 50-6-412 18.0 14.4 144 564.4 456.4 18 G4 50-6-413 18.0 14.4 144 564.4 456.4 20 50-6-415	6		50-4-411	18.0	13.7	144	389.2	317.2
9 50-4-414 18.0 13.7 144 389.2 317.2 10 50-4-415 18.0 13.7 144 389.2 317.2 11 50-6-311 18.0 14.4 96 564.4 456.4 12 50-6-312 18.0 14.4 96 564.4 456.4 13 G3 50-6-313 18.0 14.4 96 564.4 456.4 14 50-6-314 18.0 14.4 96 564.4 456.4 15 50-6-315 18.0 14.4 96 564.4 456.4 16 50-6-411 18.0 14.4 144 564.4 456.4 17 50-6-412 18.0 14.4 144 564.4 456.4 18 G4 50-6-413 18.0 14.4 144 564.4 456.4 19 50-6-413 18.0 14.4 144 564.4 456.4 20 50-6-415 18.0 <td></td> <td></td> <td>50-4-412</td> <td>18.0</td> <td>13.7</td> <td>144</td> <td>389.2</td> <td>317.2</td>			50-4-412	18.0	13.7	144	389.2	317.2
10 50-4-415 18.0 13.7 144 389.2 317.2 11 50-6-311 18.0 14.4 96 564.4 456.4 12 50-6-312 18.0 14.4 96 564.4 456.4 13 G3 50-6-313 18.0 14.4 96 564.4 456.4 14 50-6-314 18.0 14.4 96 564.4 456.4 15 50-6-315 18.0 14.4 96 564.4 456.4 16 50-6-411 18.0 14.4 144 564.4 456.4 17 50-6-412 18.0 14.4 144 564.4 456.4 18 G4 50-6-413 18.0 14.4 144 564.4 456.4 19 50-6-414 18.0 14.4 144 564.4 456.4 20 50-6-415 18.0 14.4 144 564.4 456.4 21 75-6-311 18.0 </td <td>8</td> <td>G2</td> <td>50-4-413</td> <td>18.0</td> <td>13.7</td> <td>144</td> <td>389.2</td> <td>317.2</td>	8	G2	50-4-413	18.0	13.7	144	389.2	317.2
11 50-6-311 18.0 14.4 96 564.4 456.4 12 50-6-312 18.0 14.4 96 564.4 456.4 13 G3 50-6-313 18.0 14.4 96 564.4 456.4 14 50-6-314 18.0 14.4 96 564.4 456.4 15 50-6-315 18.0 14.4 96 564.4 456.4 16 50-6-411 18.0 14.4 144 564.4 456.4 17 50-6-412 18.0 14.4 144 564.4 456.4 18 G4 50-6-413 18.0 14.4 144 564.4 456.4 19 50-6-414 18.0 14.4 144 564.4 456.4 20 50-6-415 18.0 14.4 144 564.4 456.4 21 75-6-311 18.0 20.6 96 872.7 764.7 22 G5 75-6-312 18.0 20.6 96 872.7 764.7 24 75-6-3	9		50-4-414	18.0	13.7	144	389.2	317.2
12 50-6-312 18.0 14.4 96 564.4 456.4 13 G3 50-6-313 18.0 14.4 96 564.4 456.4 14 50-6-314 18.0 14.4 96 564.4 456.4 15 50-6-315 18.0 14.4 96 564.4 456.4 16 50-6-411 18.0 14.4 144 564.4 456.4 17 50-6-412 18.0 14.4 144 564.4 456.4 18 G4 50-6-413 18.0 14.4 144 564.4 456.4 19 50-6-414 18.0 14.4 144 564.4 456.4 20 50-6-415 18.0 14.4 144 564.4 456.4 21 75-6-311 18.0 20.6 96 872.7 764.7 22 G5 75-6-312 18.0 20.6 96 872.7 764.7 24 75-6-313 18.0 20.6 96 872.7 764.7 24 75-6-3	10		50-4-415	18.0	13.7	144	389.2	317.2
13 G3 50-6-313 18.0 14.4 96 564.4 456.4 14 50-6-314 18.0 14.4 96 564.4 456.4 15 50-6-315 18.0 14.4 96 564.4 456.4 16 50-6-411 18.0 14.4 144 564.4 456.4 17 50-6-412 18.0 14.4 144 564.4 456.4 18 G4 50-6-413 18.0 14.4 144 564.4 456.4 19 50-6-414 18.0 14.4 144 564.4 456.4 20 50-6-415 18.0 14.4 144 564.4 456.4 21 75-6-311 18.0 20.6 96 872.7 764.7 22 G5 75-6-312 18.0 20.6 96 872.7 764.7 23 75-6-313 18.0 20.6 96 872.7 764.7 24 75-6-314 18.0 20.6 112 872.7 764.7 25 G6 <td>11</td> <td>1 1</td> <td>50-6-311</td> <td>18.0</td> <td>14.4</td> <td>96</td> <td>564.4</td> <td>456.4</td>	11	1 1	50-6-311	18.0	14.4	96	564.4	456.4
14 50-6-314 18.0 14.4 96 564.4 456.4 15 50-6-315 18.0 14.4 96 564.4 456.4 16 50-6-411 18.0 14.4 144 564.4 456.4 17 50-6-412 18.0 14.4 144 564.4 456.4 18 G4 50-6-413 18.0 14.4 144 564.4 456.4 19 50-6-414 18.0 14.4 144 564.4 456.4 20 50-6-415 18.0 14.4 144 564.4 456.4 21 75-6-311 18.0 20.6 96 872.7 764.7 22 G5 75-6-312 18.0 20.6 96 872.7 764.7 23 75-6-313 18.0 20.6 96 872.7 764.7 24 75-6-314 18.0 20.6 112 872.7 764.7 25 G6 75-6-315 18.0 20.6 112 872.7 764.7	12		50-6-312	18.0	14.4	96	564.4	456.4
15 50-6-315 18.0 14.4 96 564.4 456.4 16 50-6-411 18.0 14.4 144 564.4 456.4 17 50-6-412 18.0 14.4 144 564.4 456.4 18 G4 50-6-413 18.0 14.4 144 564.4 456.4 19 50-6-414 18.0 14.4 144 564.4 456.4 20 50-6-415 18.0 14.4 144 564.4 456.4 21 75-6-311 18.0 20.6 96 872.7 764.7 22 G5 75-6-312 18.0 20.6 96 872.7 764.7 23 75-6-313 18.0 20.6 96 872.7 764.7 24 75-6-314 18.0 20.6 112 872.7 764.7 25 G6 75-6-315 18.0 20.6 112 872.7 764.7	13	G3	50-6-313	18.0	14.4	96	564.4	456.4
16 50-6-411 18.0 14.4 144 564.4 456.4 17 50-6-412 18.0 14.4 144 564.4 456.4 18 G4 50-6-413 18.0 14.4 144 564.4 456.4 19 50-6-414 18.0 14.4 144 564.4 456.4 20 50-6-415 18.0 14.4 144 564.4 456.4 21 75-6-311 18.0 20.6 96 872.7 764.7 22 G5 75-6-312 18.0 20.6 96 872.7 764.7 23 75-6-313 18.0 20.6 96 872.7 764.7 24 75-6-314 18.0 20.6 112 872.7 764.7 25 G6 75-6-315 18.0 20.6 112 872.7 764.7	14		50-6-314	18.0	14.4	96	564.4	456.4
17 50-6-412 18.0 14.4 144 564.4 456.4 18 G4 50-6-413 18.0 14.4 144 564.4 456.4 19 50-6-414 18.0 14.4 144 564.4 456.4 20 50-6-415 18.0 14.4 144 564.4 456.4 21 75-6-311 18.0 20.6 96 872.7 764.7 22 G5 75-6-312 18.0 20.6 96 872.7 764.7 23 75-6-313 18.0 20.6 96 872.7 764.7 24 75-6-314 18.0 20.6 112 872.7 764.7 25 G6 75-6-315 18.0 20.6 112 872.7 764.7	15		50-6-315	18.0	14.4	96	564.4	456.4
18 G4 50-6-413 18.0 14.4 144 564.4 456.4 19 50-6-414 18.0 14.4 144 564.4 456.4 20 50-6-415 18.0 14.4 144 564.4 456.4 21 75-6-311 18.0 20.6 96 872.7 764.7 22 G5 75-6-312 18.0 20.6 96 872.7 764.7 23 75-6-313 18.0 20.6 96 872.7 764.7 24 75-6-314 18.0 20.6 112 872.7 764.7 25 G6 75-6-315 18.0 20.6 112 872.7 764.7	16		50-6-411	18.0	14.4	144	564.4	456.4
19 50-6-414 18.0 14.4 144 564.4 456.4 20 50-6-415 18.0 14.4 144 564.4 456.4 21 75-6-311 18.0 20.6 96 872.7 764.7 22 G5 75-6-312 18.0 20.6 96 872.7 764.7 23 75-6-313 18.0 20.6 96 872.7 764.7 24 75-6-314 18.0 20.6 112 872.7 764.7 25 G6 75-6-315 18.0 20.6 112 872.7 764.7	17		50-6-412	18.0	14.4	144	564.4	456.4
20 50-6-415 18.0 14.4 144 564.4 456.4 21 75-6-311 18.0 20.6 96 872.7 764.7 22 G5 75-6-312 18.0 20.6 96 872.7 764.7 23 75-6-313 18.0 20.6 96 872.7 764.7 24 75-6-314 18.0 20.6 112 872.7 764.7 25 G6 75-6-315 18.0 20.6 112 872.7 764.7	18	G4	50-6-413	18.0	14.4	144	564.4	456.4
21 75-6-311 18.0 20.6 96 872.7 764.7 22 G5 75-6-312 18.0 20.6 96 872.7 764.7 23 75-6-313 18.0 20.6 96 872.7 764.7 24 75-6-314 18.0 20.6 112 872.7 764.7 25 G6 75-6-315 18.0 20.6 112 872.7 764.7	19		50-6-414	18.0	14.4	144	564.4	456.4
22 G5 75-6-312 18.0 20.6 96 872.7 764.7 23 75-6-313 18.0 20.6 96 872.7 764.7 24 75-6-314 18.0 20.6 112 872.7 764.7 25 G6 75-6-315 18.0 20.6 112 872.7 764.7	20		50-6-415	18.0	14.4	144	564.4	456.4
23 75-6-313 18.0 20.6 96 872.7 764.7 24 75-6-314 18.0 20.6 112 872.7 764.7 25 G6 75-6-315 18.0 20.6 112 872.7 764.7	21	77	75-6-311	18.0	20.6	96	872.7	764.7
24 75-6-314 18.0 20.6 112 872.7 764.7 25 G6 75-6-315 18.0 20.6 112 872.7 764.7	22	G5	75-6-312	18.0	20.6	96	872.7	764.7
25 G6 75-6-315 18.0 20.6 112 872.7 764.7	23		75-6-313	18.0	20.6	96	872.7	764.7
	24		75-6-314	18.0	20.6	112	872.7	764.7
	25	G6	75-6-315	18.0	20.6	112	872.7	764.7
			75-6-316	18.0	20.6	112	872.7	764.7

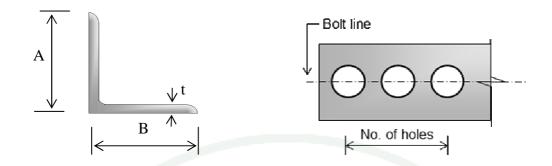


Figure 11 Definition of bolt line, no. of holes and dimension A, B and t.



Figure 12 Definition of \tilde{x} , L and d

The general form of angle denomination which is shown in Table 2 is defined by "A-T-HNS". Each character at a certain position is explained as follows;

- First position (A), represents the width of angle legs in millimeters. 50 = 50x50, 75 = 75x75 and 100 = 100x100.
- Second position (T), represents the nominal thickness of angle legs in millimeters. 4 mm, 6 mm and 7 mm.
 - Third position (H), represents the number of holes in 1 line.
 - Fourth position (N), represents the number of lines.
 - Fifth position (S), represents the number of specimens

Table 3 Description of gusset plates

	Gusset plates	A	В	t	Number		d	L
No.	denomination	(mm)	(mm)	(mm)	of bolt lines	holes per line	(mm)	(mm)
1	100-10-31B	100	550	10	1	3	18	96
2	100-10-41B	100	550	10	1	4	18	144
3	100-10-31C	100	550	10	1	3	18	102

The general form of gusset plate denomination which is shown in Table 3 is defined by "A-T-HND". Each character at a certain position is explained as follows;

- First position (A), represents the length of gusset plate in millimeters. $100 = 100 \times 550$.
- Second position (T), represents the nominal thickness of gusset plate in millimeters.
 - Third position (H), represents the number of holes in 1 line.
 - Fourth position (N), represents the number of lines.
- Fifth position (D), represents the spacing between holes. $B=48.0\ mm$, $C=56.0\ mm$.

Test set-up

- 1. Prepare the test specimen as shown in Figure 13, 14, 15
 - Align an angle to the 2 gusset plates
 - Snug-tight (lightly tighten) the bolts to the prepared holes
- Attach 2 strain gages to the specimen. Each is installed on the angle near the innermost hole of each end. Make sure that the washers do not touch the strain gages.
- 2. Set up the prepared specimen to the Universal Testing Machine (UTM). Make sure that the specimen is vertically aligned so that effects of bending will not occur (Figure 14).
- 3. Connect strain gages SG1 and SG2 electric wires to the Data Logger (Figure 13)
- 4. Apply 2 kN to the system so that the bolts have a full bearing face to the connected parts. This method can solve the problem of major slip at connection.
 - 5. Set UTM tensile load to zero.
- 6. Apply force until some failure mode is shown. In this case, tearing at the holes is expected.
 - 7. Collect data from Data Logger and plot graphs.

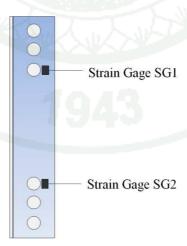


Figure 13 Strain gage positions

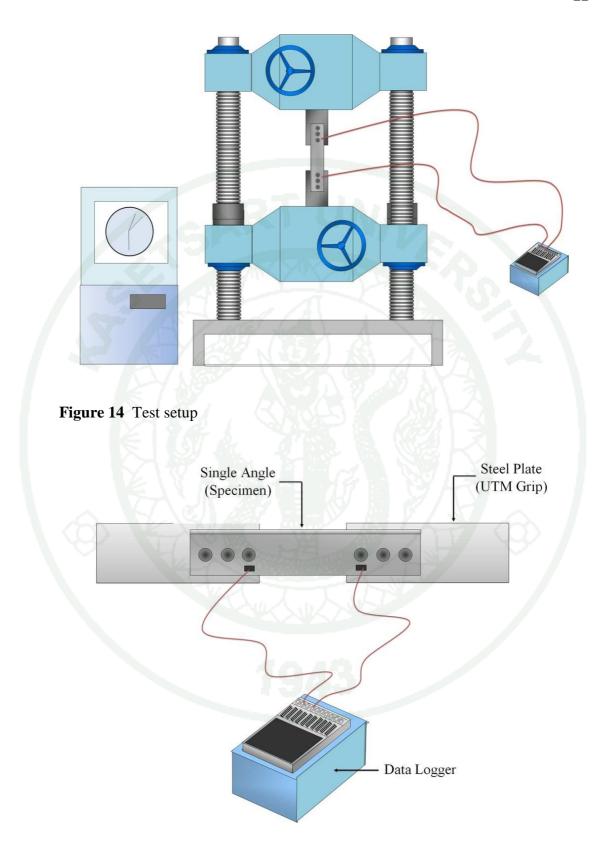


Figure 15 Data logger setup

From AISC (2010), Chapter D Section D.2

$$P_n = F_u A_e$$

(10)

with
$$A_e = UA_n$$

Notation and definitions of the terms are;

 A_e = effective net area, (mm²)

 $A_n = net area, (mm^2)$

 F_u = specified minimum tensile strength of the type of steel being used, MPa

U = the shear lag factor.

The Equation 10 can be rewritten as,

$$U = \frac{P_n}{F_u A_n}$$
(11)

And

$$U_{exp} = \frac{P_{exp}}{F_u A_n}$$
(12)

Notation and definitions of the terms are;

 $U_{exp} =$ Shear lag factor from experimental test.

 P_{exp} = Ultimate tensile load from experimental test.



Figure 16 Angle section 50x50x4THK., 500mm length



Figure 17 Angle section 50x50x6THK., 500mm length.



Figure 18 Angle section 75x75x6THK., 500mm length.



Figure 19 Strain gages attachment

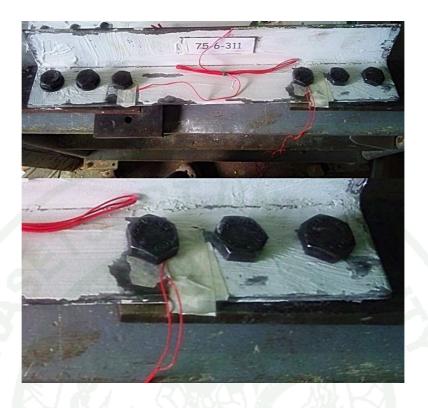


Figure 20 Strain gage attached position



Figure 21 Specimen installation

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Figure 22 Specimen installation



Figure 23 Test process



Figure 24 Collecting data via Data Logger

RESULT AND DISCUSSION

The shear lag test result of 26 specimens is recorded as shown in Table 4.

where,

 P_n = Nominal tensile strength, (kN)

 P_{exp} = Ultimate tensile load from experimental test, (kN)

 F_{u1} = minimum calculated tensile strength of the type of steel being used, (400MPa)

 $F_{u2} = \text{minimum calculated tensile strength of the type of steel being used,}$ (455MPa)

 $F_{u3} = \text{minimum calculated tensile strength of the type of steel being used,}$ (510MPa)

A = gross area of steel angles without bolt holes, (mm²)

 A_n = net area of steel angles, (mm²)

 $\begin{table 4.5cm} \textbf{Table 4} & Comparison between P_{exp} and P_n based on shear lag factor from AISC Case \\ & 2 \ and \ Case \ 8 \end{table}$

				Case2			Case 8	
		Pexp	P _n					
Test	Angle		$[F_{u1}]$	$[F_{u2}]$	$[F_{u3}]$	$[F_{u1}]$	$[F_{u2}]$	$[F_{u3}]$
		(kN)	(kN)	(kN)	(kN)	(kN)	(kN)	(kN)
1	50-4-311	104.5	108.8	123.7	138.7	76.1	86.6	97.1
2	50-4-312	100.3	108.8	123.7	138.7	76.1	86.6	97.1
3	50-4-313	97.0	108.8	123.7	138.7	76.1	86.6	97.1
4	50-4-314	93.6	108.8	123.7	138.7	76.1	86.6	97.1
5	50-4-315	102.3	108.8	123.7	138.7	76.1	86.6	97.1
6	50-4-411	123.8	114.8	130.6	146.4	101.5	115.5	129.4
7	50-4-412	121.8	114.8	130.6	146.4	101.5	115.5	129.4
8	50-4-413	121.6	114.8	130.6	146.4	101.5	115.5	129.4
9	50-4-414	120.8	114.8	130.6	146.4	101.5	115.5	129.4
10	50-4-415	125.6	114.8	130.6	146.4	101.5	115.5	129.4
11	50-6-311	157.5	155.2	176.5	197.8	109.5	124.6	139.7
12	50-6-312	153.0	155.2	176.5	197.8	109.5	124.6	139.7
13	50-6-313	147.1	155.2	176.5	197.8	109.5	124.6	139.7
14	50-6-314	150.1	155.2	176.5	197.8	109.5	124.6	139.7
15	50-6-315	146.4	155.2	176.5	197.8	109.5	124.6	139.7
16	50-6-411	167.8	164.3	186.9	209.5	146.0	166.1	186.2
17	50-6-412	167.0	164.3	186.9	209.5	146.0	166.1	186.2
18	50-6-413	180.3	164.3	186.9	209.5	146.0	166.1	186.2
19	50-6-414	160.1	164.3	186.9	209.5	146.0	166.1	186.2
20	50-6-415	164.7	164.3	186.9	209.5	146.0	166.1	186.2
21	75-6-311	235.4	240.2	273.3	306.3	183.5	208.8	234.0
22	75-6-312	274.7	240.2	273.3	306.3	183.5	208.8	234.0
23	75-6-313	269.8	240.2	273.3	306.3	183.5	208.8	234.0
24	75-6-314	240.3	249.6	283.9	318.3	183.5	208.8	234.0
25	75-6-315	238.4	249.6	283.9	318.3	183.5	208.8	234.0
26	75-6-316	288.4	249.6	283.9	318.3	183.5	208.8	234.0

				Case 2		Case 8			
	. 1	D	$0.75P_{n}$	$0.75P_{n}$	0.75P _n	0.75P _n	0.75P _n	0.75P _n	
Test	Angle	P_{exp}	$[F_{u1}]$	$[F_{u2}]$	$[F_{u3}]$	$[F_{u1}]$	$[F_{u2}]$	$[F_{u3}]$	
		(kN)	(kN)	(kN)	(kN)	(kN)	(kN)	(kN)	
1	50-4-311	104.5	81.6	92.8	104.0	57.1	64.9	72.8	
2	50-4-312	100.3	81.6	92.8	104.0	57.1	64.9	72.8	
3	50-4-313	97.0	81.6	92.8	104.0	57.1	64.9	72.8	
4	50-4-314	93.6	81.6	92.8	104.0	57.1	64.9	72.8	
5	50-4-315	102.3	81.6	92.8	104.0	57.1	64.9	72.8	
6	50-4-411	123.8	86.1	97.9	109.8	76.1	86.6	97.1	
7	50-4-412	121.8	86.1	97.9	109.8	76.1	86.6	97.1	
8	50-4-413	121.6	86.1	97.9	109.8	76.1	86.6	97.1	
9	50-4-414	120.8	86.1	97.9	109.8	76.1	86.6	97.1	
10	50-4-415	125.6	86.1	97.9	109.8	76.1	86.6	97.1	
11	50-6-311	157.5	116.4	132.4	148.4	82.2	93.4	104.7	
12	50-6-312	153.0	116.4	132.4	148.4	82.2	93.4	104.7	
13	50-6-313	147.1	116.4	132.4	148.4	82.2	93.4	104.7	
14	50-6-314	150.1	116.4	132.4	148.4	82.2	93.4	104.7	
15	50-6-315	146.4	116.4	132.4	148.4	82.2	93.4	104.7	
16	50-6-411	167.8	123.2	140.2	157.1	109.5	124.6	139.7	
17	50-6-412	167.0	123.2	140.2	157.1	109.5	124.6	139.7	
18	50-6-413	180.3	123.2	140.2	157.1	109.5	124.6	139.7	
19	50-6-414	160.1	123.2	140.2	157.1	109.5	124.6	139.7	
20	50-6-415	164.7	123.2	140.2	157.1	109.5	124.6	139.7	
21	75-6-311	235.4	180.2	205.0	229.7	137.6	156.6	175.5	
22	75-6-312	274.7	180.2	205.0	229.7	137.6	156.6	175.5	
23	75-6-313	269.8	180.2	205.0	229.7	137.6	156.6	175.5	
24	75-6-314	240.3	187.2	213.0	238.7	137.6	156.6	175.5	
25	75-6-315	238.4	187.2	213.0	238.7	137.6	156.6	175.5	
26	75-6-316	288.4	187.2	213.0	238.7	137.6	156.6	175.5	

Table 6 Comparison of shear lag factor from AISC 2010 and experimental program

				$U_{exp} = P_{exp} / F_u A_n$			
		U_{AISC}	U_{AISC}	U _{exp}	U _{exp}	U _{exp}	
Test	Angle	2010	2010	$[F_{u1}]$	$[F_{u2}]$	$[F_{u3}]$	
		Case 2,	Case 8	400	455	510	
		$(1-x^{\sim}/L)$	Case o	MPa	MPa	MPa	
1	50-4-311	0.857	0.6	0.824	0.724	0.646	
2	50-4-312	0.857	0.6	0.790	0.695	0.620	
3	50-4-313	0.857	0.6	0.765	0.672	0.600	
4	50-4-314	0.857	0.6	0.737	0.648	0.578	
5	50-4-315	0.857	0.6	0.807	0.709	0.633	
6	50-4-411	0.905	0.8	0.975	0.858	0.765	
7	50-4-412	0.905	0.8	0.960	0.844	0.753	
8	50-4-413	0.905	0.8	0.959	0.843	0.752	
9	50-4-414	0.905	0.8	0.952	0.837	0.747	
10	50-4-415	0.905	0.8	0.990	0.870	0.777	
11	50-6-311	0.850	0.6	0.863	0.759	0.677	
12	50-6-312	0.850	0.6	0.838	0.737	0.657	
13	50-6-313	0.850	0.6	0.806	0.708	0.632	
14	50-6-314	0.850	0.6	0.822	0.723	0.645	
15	50-6-315	0.850	0.6	0.802	0.705	0.629	
16	50-6-411	0.900	0.8	0.919	0.808	0.721	
17	50-6-412	0.900	0.8	0.915	0.804	0.718	
18	50-6-413	0.900	0.8	0.987	0.868	0.774	
19	50-6-414	0.900	0.8	0.877	0.771	0.688	
20	50-6-415	0.900	0.8	0.902	0.793	0.708	
21	75-6-311	0.785	0.6	0.770	0.677	0.604	
22	75-6-312	0.785	0.6	0.898	0.789	0.704	
23	75-6-313	0.785	0.6	0.882	0.775	0.692	
24	75-6-314	0.816	0.6	0.786	0.691	0.616	
25	75-6-315	0.816	0.6	0.779	0.685	0.611	
26	75-6-316	0.816	0.6	0.943	0.829	0.740	

Result discussion

American Institute of Steel Construction (AISC, 2010) had proposed a shear lag factor of single angles as shown in Table 1:

AISC 2010 proposed two different approaches for shear lag effect calculation. AISC 2010 case 2 gave a formula as shown in Table 1, while there were two constant factors to be applied when considering the shear lag effects by Case 8. However, Case 8 also proposed additional option to use the greater value between both cases.

Shear lag factors calculated by Case 2 approach always give greater values than Case 8. For instance, single angle with 3 fasteners has shear lag factor around 0.8 according to AISC case 2, but it is 0.6 if considered by AISC case 8. 30% difference is unfavorable.

Table 7 Comparison of shear lag factor between AISC 2010, Case 2 and Case 8

Test	Angle	U_{AISC} 2010	$\begin{array}{c} U_{AISC} \\ 2010 \end{array}$	(U _{AISC,Case8} - U _{AISC,Case2}) x 100 / (U _{AISC,Case2})
Test	ringic	Case 2, (1-x~/L)	Case 8	(%)
1	50-4-311	0.857	0.6	-30.0
2	50-4-312	0.857	0.6	-30.0
3	50-4-313	0.857	0.6	-30.0
4	50-4-314	0.857	0.6	-30.0
5	50-4-315	0.857	0.6	-30.0
6	50-4-411	0.905	0.8	-11.6
7	50-4-412	0.905	0.8	-11.6
8	50-4-413	0.905	0.8	-11.6
9	50-4-414	0.905	0.8	-11.6
10	50-4-415	0.905	0.8	-11.6
11	50-6-311	0.850	0.6	-29.4
12	50-6-312	0.850	0.6	-29.4
13	50-6-313	0.850	0.6	-29.4
14	50-6-314	0.850	0.6	-29.4
15	50-6-315	0.850	0.6	-29.4
16	50-6-411	0.900	0.8	-11.1
17	50-6-412	0.900	0.8	-11.1
18	50-6-413	0.900	0.8	-11.1
19	50-6-414	0.900	0.8	-11.1
20	50-6-415	0.900	0.8	¥11.1
21	75-6-311	0.785	0.6	-23.6
22	75-6-312	0.785	0.6	-23.6
23	75-6-313	0.785	0.6	-23.6
24	75-6-314	0.816	0.6	-26.5
25	75-6-315	0.816	0.6	-26.5
26	75-6-316	0.816	0.6	-26.5

1. Effect of number of bolts

For the angle specimens with similar connection length, ultimate tensile strength and size/cross section, the angle specimens with 4 bolts give ~10% to ~19% higher shear lag factor when compared to the angle specimens with 3.

2. Effect of connection length

For the angle specimens with similar numbers of bolted connections, ultimate tensile strength and size/cross section, the angle specimens with more connection length give ~4% higher shear lag factor.

3. Effect of size/cross section

For the angle specimens with similar numbers of bolted connections, ultimate tensile strength and connection length, the angle specimens with smaller size/cross section give higher shear lag factor. By 31% increase of net section angle, the shear lag factor values reduce \sim 4% to \sim 5%.

4. Effect of ultimate tensile strength of material

For the angle specimens with similar numbers of bolted connections, size/cross section and connection length, the angle specimens with lower ultimate tensile strength give higher shear lag factor.

From the test results (Table 5), the change of ultimate tensile strength 400 MPa, 455 MPa and 510 MPa is one of the major changes of shear lag factor. Table 8 shows the comparison of U_{exp} between U_{AISC} , Case 2 and U_{AISC} , Case 8.

Table 8 Comparison of U_{exp} with $U_{AISC,\,2010}$ Case 2 and $U_{AISC,2010}$ Case 8

		Compa	re with U	J _{AISC, 2010}	Compa	are with	UAISC, 2010
T4	Angle	1	Case 2	11150, 2010	Case 8		
Test	denomination	U _{exp}	U _{exp}	U _{exp}	U _{exp}	U _{exp}	U _{exp}
	denomination	$[F_{u1}]$	$[F_{u2}]$	$[F_{u3}]$	$[F_{u1}]$	$[F_{u2}]$	$[F_{u3}]$
1	50-4-311	L	L	L	Н	Н	Н
2	50-4-312	L	L	L	Н	Н	Н
3	50-4-313	L	L	L	Н	Н	E
4	50-4-314	L	L	L	Н	H	L
5	50-4-315	L	L	L	Н	Н	Н
6	50-4-411	Н	L	L	Н	Н	L
7	50-4-412	H	L	L	Н	Н	L
8	50-4-413	Н	L	L	Н	Н	L
9	50-4-414	Н	L	L	H	Н	L
10	50-4-415	H /	L	L	Н	Н	L
11	50-6-311	H	L	L	H	Н	H
12	50-6-312	L	L	L	H	H	Н
13	50-6-313	L	L	L	Н	H	Н
14	50-6-314	L	L	L	Н	Н	Н
15	50-6-315	L	L	L	Н	Н	Н
16	50-6-411	Н	L	L	Н	Н	L
17	50-6-412	Н	L	L	Н	Н	L
18	50-6-413	Н	L	L	Н	Н	L
19	50-6-414	L	- L	L	H	L	L
20	50-6-415	Н	L	L	H	L	L
21	75-6-311	L	L	L	Н	Н	Н
22	75-6-312	Н	Н	L	Н	Н	Н
23	75-6-313	Н	L	L	Н	Н	H
24	75-6-314	L	\L\	L	Н	Н	Н
25	75-6-315	L	L	L	Н	Н	Н
26	75-6-316	Н	Н	L	Н	Н	Н

The meaning of "H" in Table 8 is the value of shear lag which is got from experimental program higher than the shear lag which is defined from AISC (2010), "L" means the value of shear lag from experimental program lower than the value of AISC (2010) and "E" means both values are equal.

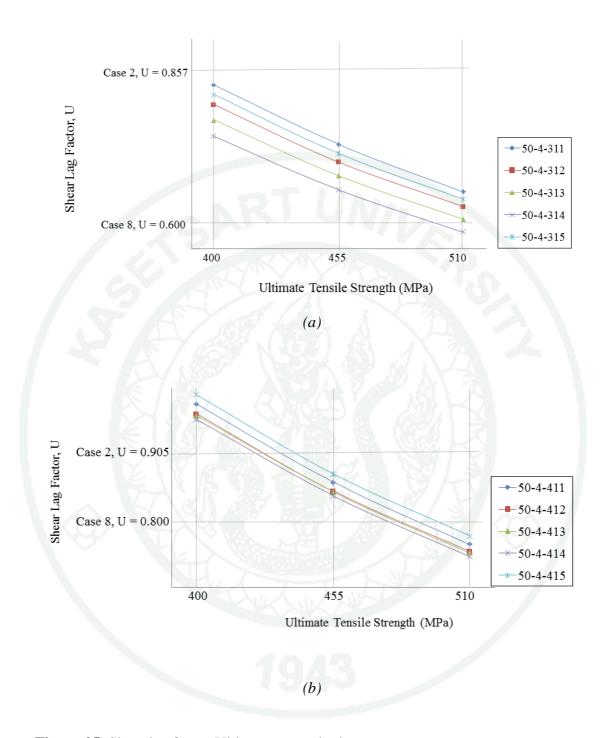


Figure 25 Shear lag factor-Ultimate strength plots

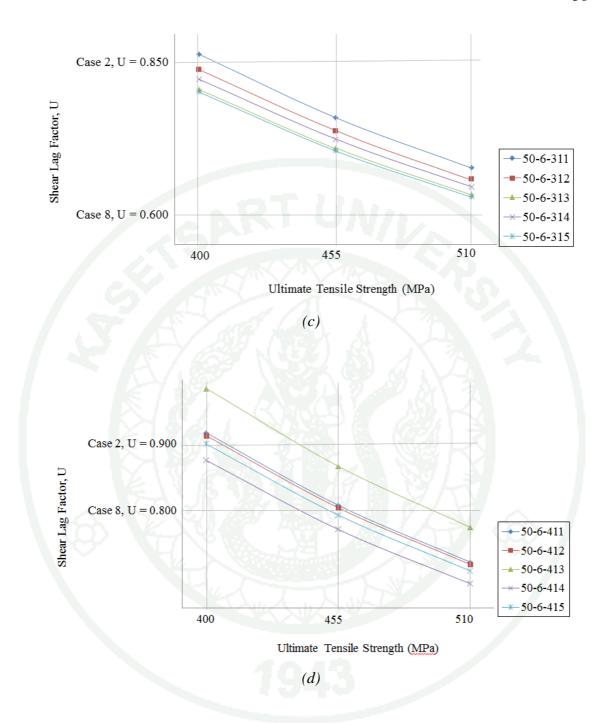


Figure 25 (Continued).

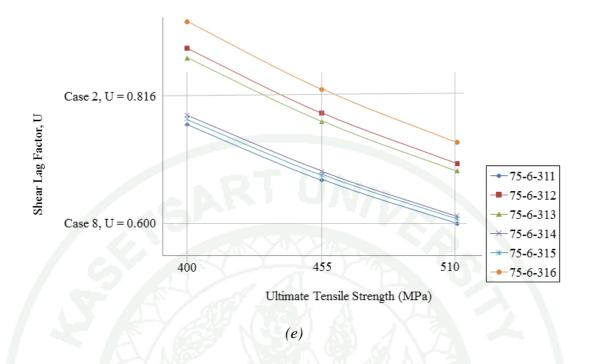


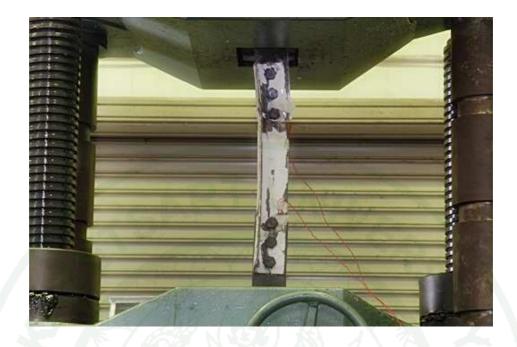
Figure 25 (Continued).

From Table 8 and Figure 25, it is obvious that, AISC 2010 Case 2 still gives safe design (Table 5, show P_{exp} and $0.75P_n$). It is concluded that 400 MPa should be selected because it always gives safe design. Therefore AISC (2010) is correct for shear lag factor (U).



(a) Failure of specimen tested

Figure 26 Experimental results.

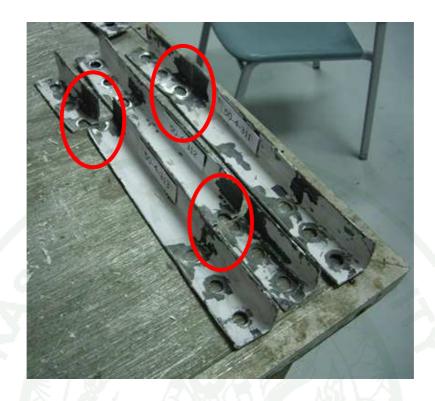


(b) Unavoidable bending



(c) Bent gusset plate during the test

Figure 26 (Continued).



(d) Failure of the specimens



(e) Failure of the specimens

Figure 26 (Continued).

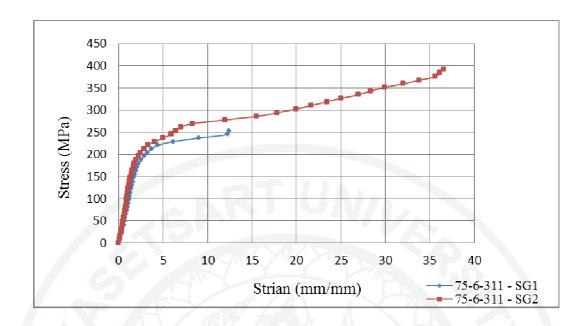


Figure 27 Stress-Strain curve plots from 75-6-311 specimen tested

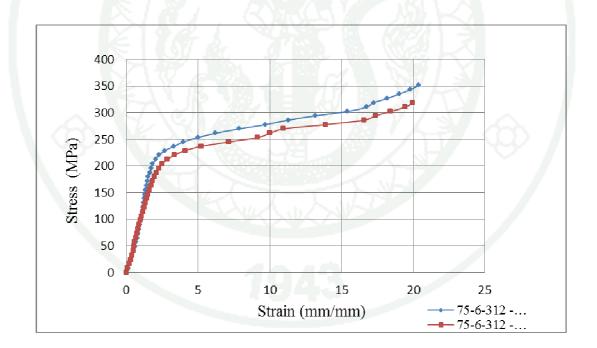


Figure 28 Stress-Strain curve plots from 75-6-312 specimen tested

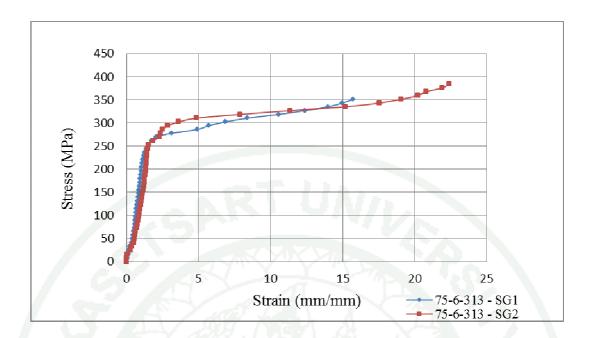


Figure 29 Axial force – Displacement plots from 75-6-313 specimen tested

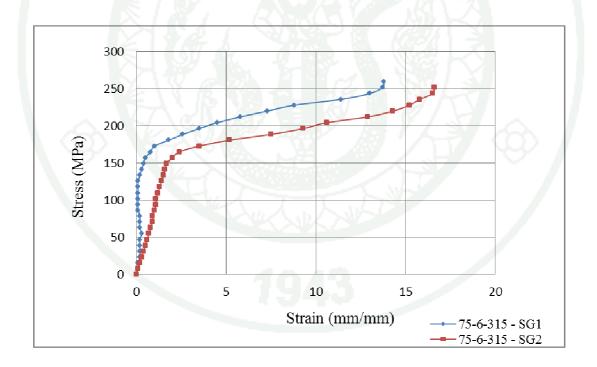


Figure 30 Stress-Strain curve plots from 75-6-315 specimen tested

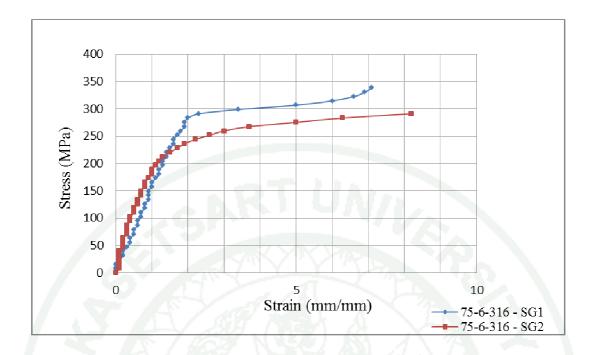


Figure 31 Stress-Strain curve plots from 75-6-316 specimen tested

CONCLUSION AND RECOMMENDATION

- 1. Shear lag effect depends on number of bolts, connection length, size/cross section of steel angles and ultimate tensile strength of material.
- 2. The shear lag factor resulted from the experiment are within the range between the values provided by AISC Case 2 to Case 8 depended on the values of ultimate tensile strength (F_u). The use of reduction factor given by AISC 2010, Case 8 seems to be conservative and recommended for higher ultimate tensile strength (F_u).
- 3. Properly chosen value of ultimate tensile strength (F_u) is important. Choosing the smallest value of F_u is always safe and recommended for designer. Further study should be focused on how to pick up the F_u value for economical design.
- 4. The strength reduction factor "φ" of 0.75 is necessary to compute the tensile strength of angle section for safety reason.

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 ${\tt 16.1-28} \qquad \qquad {\tt BUILT-UP\,MEMBERS} \qquad \qquad {\tt [Sect.\,D4.}$

	Shea	TABLI r Lag Factors to Tension	s for Connection	ns		
Case	Description	of Element	Shear Lag Factor, U	Example		
1	All tension members load is transmitted di cross-sectional eleme welds (except as in C	rectly to each of the ents by fasteners or	<i>U</i> = 1.0			
2	dinal welds or by lon combination with tran natively, for W, M, S	tension load is trans- ot all of the cross- y fasteners or longitu- gitudinal welds in isverse welds, (Alter-	U= 1− X /	İT		
3	All tension members load is transmitted or welds to some but no cross-sectional eleme	nly by transverse ot all of the	U = 1.0 and A_n = area of the directly connected elements	W		
4	Plates where the tentransmitted by longitu		$l \ge 2wU = 1.0$ $2w > l \ge 1.5wU = 0.87$ $1.5w > l \ge wU = 0.75$	***		
5	Round HSS with a si gusset plate	ngle concentric	$l \ge 1.3DU = 1.0$ $D \le l < 1.3DU = 1 - \overline{x} / f$ $\overline{x} = D / \pi$			
6	Rectangular HSS	with a single concentric gusset plate	$I \ge HU = 1 - \overline{X}_{I}$ $\overline{X} = \frac{B^2 + 2BH}{4(B+H)}$	H N		
		with two side gusset plates	$I \ge HU = 1 - \overline{X}_{f}$ $\overline{X} = \frac{B^2}{4(B+H)}$	H		
7	W, M, S or HP Shapes or Tees cut from these shapes. (If U is calculated per Case 2, the	with flange con- nected with 3 or more fasteners per line in the direction of loading	$b_f \ge 2/3 dU = 0.90$ $b_f < 2/3 dU = 0.85$			
	larger value is per- mitted to be used.)	with web connected with 4 or more fas- teners per line in the direction of loading	<i>U</i> = 0.70	-		
154.00	Single and double angles (If <i>U</i> is calculated per	with 4 or more fas- teners per line in the direction of loading	<i>U</i> = 0.80	**************************************		
	Case 2, the larger value is permitted to be used.)	with 3 fasteners per line in the direction of loading (With fewer than 3 fasten- ers per line in the direction of loading, use Case 2.)	<i>U</i> = 0.60	_		

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Appendix Figure 1 Table D3.1 shear lag factor for connections to tension members AISC (2010)

JIS G3101 SS400 steel plate/sheet for general purpose structural steels

JIS G3101 SS400 steel plate/sheet, JIS G3101 SS400 steel plate/sheet, under JISG3101 standard, we can regard SS400 steel plate/sheet for general purpose structural steel.

JIS G3101 SS400 is a technical delivery conditions for general purpose structural steel.JIS G3101 SS400 is a type of steel sheet under JIS standard which is used to build ship, bridge, belongs to high strength sheet. JIS G3101 SS400 is equivalent to DIN:St37-2,EN S235JR,ASTM:A283C and UNI:FE360B.

	Con	Comparision of steel grades				
	BS 4360	40(A)B				
SS400	CSAG40-21	230 G				
JIS3101	IS	IS 226				
	JIS 3106	SM 400 A				
	ISO 630	Fe 360 B				
	ASTM	A 36/A 283 C				

Chemical Composition

Grade	Chemical Composition, % by weight								
	C. max	Si. max	Manganese	P. max	S. max				
SS400	-///	1 Rais	-\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\	0.050	0.050				

Mechnical Propertites

Grade	Yield Strength min. (Mpa)		Tensile Strength MPa	E	Impact Resistance min.[J]		
	Thickness <16 mm	Thickness ≥16mm		Thickness <5mm	Thickness 5-16mm	Thickness ≥16mm	
SS400	245	235	400-510	21	17	21	3/ 4/

Appendix Figure 2 JIS3101 Grade SS400 material properties

Appendix Table Calculation of "U" which give from experimental program

Test	Angle	$P_{\rm exp}$	A_n	F_{u1}	F_{u2}	F_{u3}	$U_{exp}\left[F_{u1}\right]$	$U_{\rm exp}\left[F_{\rm u2}\right]$	$U_{exp}\left[F_{u3}\right]$
1031	denomination	(kN)	(mm2)	(Mpa)	(Mpa)	(Mpa)	$P_{exp}/(F_{u1}.A_n)$	$P_{exp}/(F_{u2}.A_n)$	$P_{exp}/(F_{u3}.A_n)$
1	50-4-311	104.49	317.2	400	455	510	0.824	0.724	0.646
2	50-4-312	100.28	317.2	400	455	510	0.790	0.695	0.620
3	50-4-313	97.00	317.2	400	455	510	0.765	0.672	0.600
4	50-4-314	93.56	317.2	400	455	510	0.737	0.648	0.578
5	50-4-315	102.34	317.2	400	455	510	0.807	0.709	0.633
6	50-4-411	123.76	317.2	400	455	510	0.975	0.858	0.765
7	50-4-412	121.82	317.2	400	455	510	0.960	0.844	0.753
8	50-4-413	121.62	317.2	400	455	510	0.959	0.843	0.752
9	50-4-414	120.81	317.2	400	455	510	0.952	0.837	0.747
10	50-4-415	125.63	317.2	400	455	510	0.990	0.870	0.777
11	50-6-311	157.55	456.4	400	455	510	0.863	0.759	0.677
12	50-6-312	152.96	456.4	400	455	510	0.838	0.737	0.657
13	50-6-313	147.06	456.4	400	455	510	0.806	0.708	0.632
14	50-6-314	150.11	456.4	400	455	510	0.822	0.723	0.645
15	50-6-315	146.35	456.4	400	455	510	0.802	0.705	0.629
16	50-6-411	167.84	456.4	400	455	510	0.919	0.808	0.721
17	50-6-412	167.02	456.4	400	455	510	0.915	0.804	0.718
18	50-6-413	180.27	456.4	400	455	510	0.987	0.868	0.774
19	50-6-414	160.12	456.4	400	455	510	0.877	0.771	0.688
20	50-6-415	164.71	456.4	400	455	510	0.902	0.793	0.708
21	75-6-311	235.44	764.7	400	455	510	0.770	0.677	0.604
22	75-6-312	274.68	764.7	400	455	510	0.898	0.789	0.704
23	75-6-313	269.78	764.7	400	455	510	0.882	0.775	0.692
24	75-6-314	240.35	764.7	400	455	510	0.786	0.691	0.616
25	75-6-315	235.44	764.7	400	455	510	0.770	0.677	0.604
26	75-6-316	240.35	764.7	400	455	510	0.786	0.691	0.616

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